

Control System (ECE411)

Lectures 13 & 14

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Steady-State Error Analysis

Remark: For a unity feedback system ($H(s) = 1$):

$$e(t) = r(t) - c(t)$$

$$E(s) = R(s) - C(s) = R(s) - R(s)M(s) = E(s) = [1 - M(s)]R(s)$$

where $M(s)$ is the closed loop transfer function.

Thus, $e_{ss} = \lim_{s \rightarrow 0} sE(s) = \lim_{s \rightarrow 0} s[1 - M(s)]R(s)$

For a unit step $R(s) = \frac{1}{s}$, we get

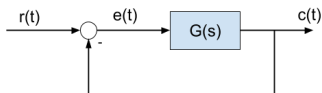
$$e_{ss} = [1 - M(0)]$$

Note: The above results could sometimes be used for cases when $H(s) \neq 1$ (*tracking error*).

Example

Given a unity feedback system shown below with closed loop transfer function

$$M(s) = \frac{K}{(s^2 + 2s + 2)(s + a)},$$



Steady-State Error Analysis-Cont.

(a) find K and a such that $e_{ss} = 1.5$ to unit ramp input,

(b) find e_{ss} for unit step input.

Part (a): First, we find the open-loop transfer function $G(s)$ from $M(s)$ using,

$$M(s) = \frac{G(s)}{1 + G(s)} = \frac{K}{(s^2 + 2s + 2)(s + a)} \implies$$

$$G(s) = \frac{K}{(s^2 + 2s + 2)(s + a) - K} = \frac{K}{s^3 + (a + 2)s^2 + (2a + 2)s + (2a - K)}$$

Now, in order to avoid a Type 0 system which yields $e_{ss} \rightarrow \infty$ for ramp input,

$$2a - K = 0 \implies K = 2a.$$

For a unit ramp input:

$$e_{ss} = \frac{1}{K_v}$$

$$\text{where } K_v = \lim_{s \rightarrow \infty} sG(s)$$

$$\text{Thus, } e_{ss} = \frac{1}{K_v} = 1.5 \implies K_v = \frac{2}{3}$$

Steady-State Error Analysis-Cont.

Using $G(s)$ in $K_v = \lim_{s \rightarrow \infty} sG(s)$ and $K_v = \frac{2}{3}$ gives,

$$K_v = \lim_{s \rightarrow \infty} \frac{K}{s^2 + (a+2)s + (2a+2)} \implies \frac{K}{2a+2} = \frac{2}{3}$$

Solving for K and a using the above equation and $K = 2a$ gives $a = 2$, and $K = 4$.

Part (b): Using the result from part (a):

$$G(s) = \frac{4}{s(s^2 + 4s + 6)}$$

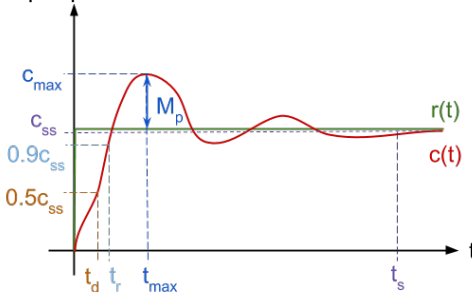
which is obviously Type 1 system $\implies e_{ss} = 0$ to unit step input.

Transient Analysis

Transient Response

Transient response allows for determining whether or not a system is stable and, if so, how stable it is (i.e. relative stability) as well as the speed of response when a step reference input is applied.

A typical time-domain response of a second order system (closed loop) to a unit step input is shown.



Transient Analysis-Cont.

Key Definitions:

- 1 **Max Overshoot (M_p)** $M_p = \frac{c_{max} - c_{ss}}{c_{ss}}$
 c_{max} : max value of $c(t)$, c_{ss} : steady-state value of $c(t)$
%max overshoot = $100 \times M_p$
 M_p determines relative stability: Large $M_p \iff$ less stable
- 2 **Delay time (t_d)**: Time for $c(t)$ to reach 50% of its final value.
- 3 **Rise time (t_r)**: Time for $c(t)$ to rise from 10% to 90% of its final value.
- 4 **Settling time (t_s)**: Time for $c(t)$ to decrease and stay within a specified (typically 5%) of c_{ss} .

Desirable characteristics: Small M_p , small t_d , quick t_r and fast t_s (cannot be accomplished simultaneously).

Transient Response of 2^{nd} -Order Control System

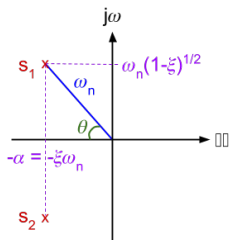
Consider a control system with closed-loop transfer function,

$$M(s) = \frac{C(s)}{R(s)} = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_n^2}, \quad M(0) = 1$$

Characteristic Equation : $\rho(s) = s^2 + 2\xi\omega_n s + \omega_n^2 = 0$
has the following roots,

$$s_{1,2} = -\xi\omega_n \pm j\omega_n \sqrt{1 - \xi^2} = -\alpha \pm j\omega$$

These are depicted in the following figure.



$$\cos \theta = -\xi, \quad \tan \theta = \frac{\sqrt{1-\xi^2}}{-\xi}$$

Transient Response of 2^{nd} -Order Control System-Cont.

Response to unit step input ($R(s) = \frac{1}{s}$) is

$$C(s) = \frac{\omega_n^2}{s \underbrace{(s^2 + 2\xi\omega_n s + \omega_n^2)}_{(s+\alpha)^2 + \omega^2}}$$

Use PFE, time-domain response is found to be

$$c(t) = \underbrace{1}_{c_{ss}} + \underbrace{\frac{e^{-\alpha t}}{\sqrt{1-\xi^2}} \sin[\omega t - \theta]}_{c_{tr}(t)}, \quad \forall t \geq 0$$

damping

$\alpha = \xi\omega_n$: Damping Factor- Controls the rate of rise time and decay time i.e. α controls damping and speed of response.

Can control oscillations by changing ω .

Can control damping by changing ξ .

Transient Response of 2^{nd} -Order Control System-Cont.

$\tau = 1/\alpha$: Time Constant

Large $\alpha \implies$ small $\tau \implies$ signal decays quickly.

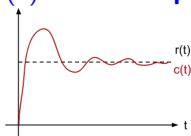
ξ : Damping Ratio (ratio between actual damping factor and the damping factor for critically damped ($\xi = 1 \implies s_{1,2} = -\omega_n$)).

ω_n : Natural Undamped Frequency ($\xi = 0 \implies s_{1,2} = \pm j\omega_n$ i.e. purely oscillatory with frequency ω_n)

$\omega = \omega_n \sqrt{1 - \xi^2}$: Conditional Frequency

Transient Response-Different Damping Cases

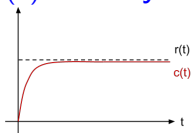
(a) Underdamped:



$$0 < \xi < 1, \quad s_{1,2} = -\xi\omega_n \pm j\omega_n\sqrt{1-\xi^2}$$

Characteristics: Small rise time (t_r), large overshoot (M_p).

(b) Critically Damped:

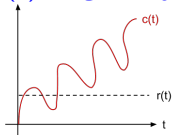


$$\xi = 1, \quad s_{1,2} = -\omega_n \text{ (repeated real roots)}$$

Characteristics: No overshoot, slow/large rise time.

Transient Response-Different Damping Cases

(e) **Negatively damped (unstable):**



$$\xi < 0, \quad s_{1,2} = -\xi\omega_n \pm j\omega_n\sqrt{1 - \xi^2}$$

Transient Response: Performance Measures

1. Peak Time (t_{max})

To find the peak time (time at which the step response reaches its maximum), we take the derivative of the step response and set it to zero.

$$\begin{aligned}\frac{dc(t)}{dt} &= 0 \\ \frac{dc(t)}{dt} &= \frac{-\xi e^{-\xi\omega_n t}}{\sqrt{1-\xi^2}} \sin[\omega t - \theta] + \frac{e^{-\xi\omega_n t}}{\sqrt{1-\xi^2}} \omega \cos[\omega t - \theta]\end{aligned}$$

Using $\omega = \omega_n \sqrt{1-\xi^2}$ and trig identities, we can simplify the above equation as:

$$\frac{dc(t)}{dt} = \frac{\omega_n}{\sqrt{1-\xi^2}} e^{-\xi\omega_n t} \sin(\omega t), \quad \forall t \geq 0,$$

Now, $\frac{dc(t)}{dt} = 0 \implies \sin(\omega t) = 0$ or when $t \rightarrow \infty$ (i.e. final value)

The first condition gives the extrema points (maxima and minima) of $c(t)$, i.e.

$$\omega t = n\pi \implies t = \frac{n\pi}{\omega_n \sqrt{1-\xi^2}}$$

The first maximum (Max overshoot) of $c(t)$ happens for $n = 1$. Thus,

$$t_{max} = \frac{\pi}{\omega_n \sqrt{1-\xi^2}}$$

Transient Response-Performance Measures-Cont.

Note: Although Max and Min of $c(t)$ occur at periodic interval, the response is NOT periodic due to damping (unless $\xi = 0$).

2. Max Overshoot (M_p)

To find M_p , we substitute t_{max} in expression for $c(t)$. This yields,

$$c_{max} = c(t) \Big|_{t=t_{max}} = 1 + e^{-\frac{\pi\xi}{\sqrt{1-\xi^2}}}$$

Thus, using the fact that $c_{ss} = 1$, we get

$$M_p = c_{max} - 1 = e^{-\frac{\pi\xi}{\sqrt{1-\xi^2}}}$$

Or in percentage,

$$\% \text{Max Overshoot} = 100e^{-\frac{\pi\xi}{\sqrt{1-\xi^2}}}$$

As can be seen, Max Overshoot is solely a function of ξ . Hence, Larger $\xi \implies$ smaller M_p ($\xi < 1$). But this would increase the delay time and rise time as seen next.

For t_d , t_r , and t_s only approximate equations can be obtained. These are given next.

Transient Response-Performance Measures-Cont.

3. Delay Time (t_d)

For t_d , we set $c(t) = 0.5$ and solve for t_d ,

$$t_d \approx \frac{1 + 0.7\xi}{\omega_n}, \quad 0 < \xi < 1$$

$t_d \approx \frac{1+0.6\xi+0.15\xi^2}{\omega_n}$: wider range of ξ and more accurate.

4. Rise Time (t_r)

$$t_r \approx \frac{0.8 + 2.5\xi}{\omega_n}, \quad 0 < \xi < 1$$

$t_r = \frac{1+1.1\xi+1.4\xi^2}{\omega_n}$, wider range of ξ and more accurate.

5. Settling Time (t_s)

$$t_s \approx 4/\xi\omega_n$$

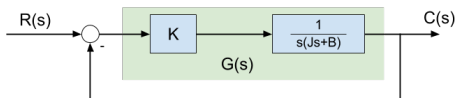
As can be seen, Small $\xi \implies$ smaller t_d and t_r but larger t_s .

Optimum Range for ξ : $0.5 \leq \xi \leq 0.8$

Analyzing Simple Controllers for 2nd Order Systems

1. Gain Controller

Consider servo control system below:



Steady State Error Analysis:

Loop transfer function:

$$G(s) = \frac{K}{s(Js+B)} \implies \text{Type 1} \implies e_{ss} = 0 \text{ for } r(t) = u_s(t)$$

For a unit ramp input $e_{ss} = \frac{1}{K_v}$ and

$$K_v = \lim_{s \rightarrow 0} sG(s)H(s) = \frac{K}{B}$$

Thus,

$$e_{ss} \text{ to unit ramp} \implies e_{ss} = \frac{B}{K}$$

which implies Small e_{ss} Requires Large Gain K .

Analyzing Simple Controllers for 2nd Order Systems-Cont.

Transient Analysis:

The closed-loop transfer function

$$M(s) = \frac{G(s)}{1+G(s)} = \frac{\frac{K}{s(sJ+B)}}{1+\frac{K}{s(Js+B)}} = \frac{K/J}{s^2+Bs/J+K/J}$$

Comparing with standard case,

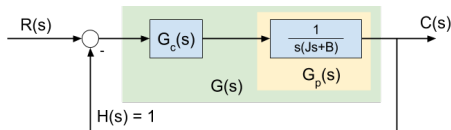
$$M(s) = \frac{\omega_n^2}{s^2+2\xi\omega_n s+\omega_n^2} \implies \omega_n = \sqrt{K/J} \text{ and } 2\omega_n\xi = B/J \implies \xi = \frac{B}{2\sqrt{KJ}}$$

Since B and J cannot be tweaked (motor parameters), Large $K \implies$ Reduced $\xi \implies$ large $M_p \implies$ i.e. less stable.

Thus, a simple gain controller (K) won't produce desirable steady-state and transient behavior as a compromise between small steady state error and good relative stability and fast response cannot be achieved.

Analyzing Simple Controllers for 2nd Order Systems-Cont.

2. Proportional-Derivative Controller (PD)



Controller transfer function $G_c(s) = K_P + K_D s$, where K_P is the proportional constant, and K_D is the derivative constant.

Steady State Error Analysis:

Loop transfer function:

$$G(s) = G_c(s)G_p(s) = \frac{K_P + K_D s}{s(Js+B)} \implies \text{Still Type 1} \implies e_{ss} = 0 \text{ for } r(t) = u_s(t)$$

For a unit ramp input $e_{ss} = \frac{1}{K_v}$ and

$$K_v = \lim_{s \rightarrow 0} sG(s)H(s) = \frac{K_P}{B} \implies e_{ss} = \frac{B}{K_P}$$

i.e. it is possible to make e_{ss} to a unit ramp as small as possible by increasing proportional Gain K_P .

Analyzing Simple Controllers for 2nd Order Systems-Cont.*Transient Analysis:*

The closed-loop transfer function

$$M(s) = \frac{G_c(s)G_p(s)}{1+G_c(s)G_p(s)} = \frac{(K_P+K_Ds)/J}{s^2 + \underbrace{\frac{(B+K_D)s}{J}}_{2\xi\omega_n} + \underbrace{\frac{K_P}{J}}_{\omega_n^2}}$$

Again, comparing with standard case,

$$M(s) = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_n^2} \implies \omega_n = \sqrt{\frac{K_P}{J}} \text{ and}$$
$$2\omega_n\xi = \frac{(B+K_D)}{J} \implies \xi = \frac{B+K_D}{2\sqrt{K_P J}}$$

Thus, we can choose:

- (a) Large K_P for small e_{ss} to unit ramp, and
- (b) Appropriate K_D to have $0.5 < \xi < 0.8$.

However, PD controller adds a zero at $s = -K_P/K_D$ which could have an impact in changing the shape of the response to unit step. Additionally, PD controller is susceptible to noise and difficult to realize.

Analyzing Simple Controllers for 2nd Order Systems-Cont.

Steady State Error Analysis:

Loop transfer function:

$$G(s)H(s) = \frac{K(1+K_t s)}{s(Js+B)} \implies \text{Still Type 1} \implies e_{ss} = 0 \text{ for } r(t) = u_s(t)$$

For a unit ramp input $e_{ss} = \frac{1}{K_v}$ and

$$K_v = \lim_{s \rightarrow 0} sG(s)H(s) = \frac{K}{B} \implies e_{ss} = \frac{B}{K}$$

i.e. it is possible to make e_{ss} to a unit ramp as small as possible by increasing Gain K .

Transient Analysis:

The closed-loop transfer function

$$M(s) = \frac{\frac{K}{J}}{s^2 + \frac{(B+K K_t)}{J}s + \frac{K}{J}}$$

Comparing with standard case,

$$M(s) = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_n^2} \implies \omega_n = \sqrt{\frac{K}{J}} \text{ and}$$

$$2\omega_n\xi = \frac{(B+K K_t)}{J} \implies \xi = \frac{B+K K_t}{2\sqrt{KJ}}$$

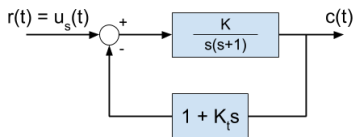
Analyzing Simple Controllers for 2nd Order Systems-Cont.

Thus, we can choose:

- (a) Large Gain K for small e_{ss} to unit ramp, and
- (b) Appropriate K_t to have $0.5 < \xi < 0.8$.

Note: Tachometer control doesn't have the same issues of the PD controller. Hence, used widely for servo control.

Example: For the tach control system below, find K and K_t such that max overshoot, M_p , to unit step is 0.2 and peak time is 1 second. Then, using these values of K and K_t , find t_r and t_s .



$$M_p = e^{-\xi\pi/\sqrt{1-\xi^2}} = 0.2 \implies \xi = 0.456$$

$$t_{max} = \frac{\pi}{\omega_n \sqrt{1-\xi^2}} = 1 \implies \omega_n = 3.53$$

Analyzing Simple Controllers for 2nd Order Systems-Cont.

But,

$$K = \omega_n^2 \implies K = 12.5$$

Also,

$$\xi = \frac{1+KK_t}{2\sqrt{K}} = 0.456 \implies K_t = 0.178$$

$$t_r = \frac{0.8+2.5\xi}{\omega_n} = 0.55\text{sec}$$

$$t_s = \frac{4}{\xi\omega_n} = 2.48\text{sec}$$